

HEAT EXCHANGER WITH CORE AND SUPPORT STRUCTURE COUPLING FOR REDUCED THERMAL STRESS

BACKGROUND

To improve the overall efficiency of a gas turbine engine, a heat exchanger or recuperator can be used to provide heated air for the turbine intake. The heat exchanger operates to transfer heat from the hot exhaust of the turbine engine to the compressed air being drawn into the turbine. As such, the turbine saves fuel it would otherwise expend raising the temperature of the intake air to the combustion temperature.

The heat of the exhaust is transferred by ducting the hot exhaust gases past the cooler intake air. Typically, the exhaust gas and the intake air ducting share multiple common walls, or other structures, which allow the heat to transfer between the two gases (or fluids depending on the specific application). That is, as the exhaust gases pass through the ducts, they heat the common walls, which in turn heat the intake air passing on the other side of the walls. Generally, the greater the surface areas of the common walls, the more heat which will transfer between the exhaust and the intake air. Also, the more heat which is transferred between the exhaust and the air, the greater the efficiency of the heat exchanger.

As shown in the cross-sectional view of Figure 1a (Fig. 1a), one example of this type of device is a heat exchanger 5, which uses a shell 10 to contain and direct the exhaust gases, and a core 20, placed within the shell 10, to contain and direct the intake air. As can be seen, the core 20 is constructed of a stack 26 of thin plates 22 which alternatively channel the inlet air and the exhaust gases through the core 20. That is, the layers 24 of the core 20 alternate between channeling the inlet air and channeling the exhaust gases. In so doing, the ducting keeps the air and exhaust gases from mixing with one another. Generally, to maximize the total heat transfer surface area of the core 20, many closely spaced plates 22 are used to define a multitude of layers 24. Further, each plate 22 is very thin and made of a material

with good mechanical and heat conducting properties. Keeping the plates 22 thin assists in the heat transfer between the hot exhaust gases and the colder inlet air.

Typically, during construction of such a heat exchanger 5, the plates 22 are positioned on top of one another and then compressed to form the stack 26. Since the plates 22 can separate if not held together, the compression of the plates 22 ensures that there are always positive compressive forces on the core 20 to hold the plates 22 in place.

Applying a high pre-load to the stack 26 reduces the potential for separation of the plates 22. However, to be able to apply pre-loads to the stack 26, a pre-load assembly or support structure 50 positioned about the stack 26, is needed. In addition to applying the pre-load to the stack 26, the support structure 50 carries any additional loading exerted by the stack 26. Such additional loads can come from a variety of sources, including thermal expansion of the stack 26 and the pressurization of air (or other medium) in the stack 26.

The support structure 50 collectively includes strongbacks 40, tie rods 30, and the shell 10. The tie rods 30 are held to the strongbacks 40 by fasteners 36 positioned at the ends 32 of the tie rods 30. Because the support structure 50 supports the core 20 (namely the stack 26) and is not a heat transfer medium, the components of the support structure 50 are made of much thicker materials than those of the core 20. Unfortunately, these thicker materials cause the support structure 50 to thermally expand at a much slower rate than the quick responding core 20, with its thin plates 22. The thickness (and thus the thermal response) of the support structure 50 will also be affected by the amount of the pre-load applied to the core 20.

Differential expansion or contraction between the core 20 and the support structure 50 can result from a variety of sources, including differential thermal expansion rates and air (or fluid) pressure variations. Differential expansion or contraction between elements of the heat exchanger 5 can occur in any dimension, and typically in all dimensions at the same time. That is, not only will the core 20 expand or contract along its length, L_{A1} , quicker than the support structure 50 will, but it also deforms faster along its width, W_{A1} , and depth (not show).

As can be seen in Figure 1a, to bring air into the core 20, an air inlet tube 23 is positioned within an inlet manifold 25. Likewise, an air outlet tube 29 is positioned within an

outlet manifold 27. However, as the core 20 expands, or contracts, along its width (and depth) faster than that of the support structure 50, the inlet manifold 25 and outlet manifold 27 will move, as shown in Figure 1b (Fig. 1b) (showing the core 20 differentially expanded). With the core 20 expanded (to a width of W_{A2}), the inlet manifold 25 and the outlet manifold 27 are no longer aligned with the respective openings of support structure 50. The misalignment of the manifolds places stresses on the tubes 23 and 29, and may result in the tubes being deformed (as shown), displaced and/or otherwise damaged. Movement of the tubes 23 and 29 may cause them to contact and damage the interior portions of the core 20. Damage to the core 20 and the tubes 23 and 25 can be costly and time consuming to correct. Further, deformation of the tubes 23 and 25, can result in a disruption and a reduction of the airflow through the core 20, which in turn, can lower the efficiency of the heat exchanger 5. Also, a reduction of the air passing through the core 20, may cause severe damage to the core 20 due to overheating.

Approaches to preventing damage from lateral expansion of the core 20 have included attempts to restrain the expansion and/or contraction of the core by application of additional compressive forces. However, such expansion/contraction restraining has resulted in the core and the support structure being put under excessive loading. This loading can result in high stresses and thermal damage or failure to both the core and the support structure. Such thermal damage includes creep and/or buckling of the associated structures.

While the structures of the heat exchanger can be enlarged to carry greater loads, doing so results in certain disadvantages. These disadvantages include: a lowering of the heat transfer characteristics of the core, an increase in the differential expansion/contraction between the core and the support structure, and an increase in the cost and weight of the heat exchanger.

One approach to accommodate the width-wise differential thermal expansion and contraction has been to use an inlet bellows 60 and an outlet bellows 70, as shown in Figure 2 (Fig. 2). The inlet bellows 60 and the outlet bellows 70 are used to keep the inlet tube 23 connected to an external inlet duct and the outlet tube 29 connected to an external outlet duct as the core 20 moves relative to the support structure 50. As the core 20 expands in width, the inlet bellows 60 and the outlet bellows 70 both deform to maintain pathways for the flow of air.

This prevents stresses from being placed on the tubes 23 and 29, as well as on the core 20.

One problem with the use of the bellows is that the outlet bellows 70 is very expensive and difficult to manufacture. This is because the outlet bellows 70 must be able function under the extreme temperatures associated with the outlet side of the core 20. Typically, these temperatures are very close to, or the same as, the temperature of exhaust gases, which enter the core 20 just after exiting from the attached turbine engine (not shown). Materials which can withstand these temperatures and continue to be sufficiently flexible over time are very expensive and difficult to use in fabricating the outlet bellows 70.

An additional problem with using a bellows system such as that shown in Figure 2, is that with repeated thermal cycling, the core 20 can migrate about relative to the support structure 50. This can result in restrictions in the airflow, damage to the bellows, and/or failure of one or both of the bellows. Also, such core movement requires that the length of the bellows be increased, which in turn increases the cost of the heat exchanger.

Therefore, a need exists for a heat exchanger which accommodates differential expansion or contraction between the core and the supporting structure, such that the airflow through the core is not significantly disrupted. The heat exchanger must be configured to prevent failures or damage caused by buckling, creep or any other similar source. Further, the heat exchanger should be relatively simple in construction and operation to minimize its cost, weight and complexity.

SUMMARY

The present invention provides a heat exchanger which in at least some embodiments includes a core, a support structure and a mount. The core is variable in its size and has a first port and a second port. The mount is positioned between the core and the support structure, adjacent to the second port of the core. The mount restrains the core relative to the support structure, such that when the core varies in size it does so either away from or towards the mount.

The heat exchanger can also include a deformable or flexible connector (e.g. bellows). This connector is attached to the core in a manner which allows it and the first port of the

core to remain in fluid communication as the core varies in size. In this manner, the heat exchanger can remain attached to a substantially fixed structure (e.g. external ducting), while the core expands and contracts. The connector can be a bellows, a flexible high temperature hose or the like.

The mount includes a pin and a receiver. The receiver receives the pin so as to restrain the movement of the core. The arrangement of the mount varies by embodiments of the invention. For example, the pin can be attached to the support structure and the receiver is defined in the core. Another embodiment has the pin attached to the core and receiver is defined in the support structure. In yet another embodiment, the mount has a core receiver, a support structure receiver and a pin. In turn, the pin has a first end and an opposing second end, with the core receiver receiving the first end and the support structure receiver receiving the second end. The core receiver is defined in the core and the support structure receiver is defined in the support structure.

In some embodiments of the present invention, a lower temperature fluid (e.g. air) passes through the first port of the core and a higher-temperature fluid (e.g. air) passes through the second port of the core. In this manner, the connector carries a lower temperature fluid and the mount is positioned adjacent the second port, which channels a high temperature fluid. As such, the connector needs only to be fabricated to carry lower temperature fluids and a minimum amount of core expansion will occur at the second port. The connector can be flexible to accommodate the expansion and contraction of the core and remain in fluid communication with the first port and any attached external fluid transport means (e.g. ducting).

In other embodiments, the heat exchanger includes a laterally expandable core, a support structure, a mount and a bellows. Being expandable, the core is variable in its size. The core has a lower temperature fluid port and a higher temperature fluid port. The mount is positioned between the core and the support structure, adjacent the higher temperature fluid port. The mount functions to restrain the core, such that the core varies in size laterally away from and towards the mount. The bellows is attached at the lower temperature fluid port. This is done so that the bellows, the lower temperature fluid port and any external ducting (e.g. tube), are in constant fluid communication as the core varies in size.

BRIEF SUMMARY OF THE DRAWINGS

Fig. 1a and b are perspective views of cross-sections of a heat exchanger and a portion of a heat exchanger.

Fig. 2 is a perspective view of a cross-section of a heat exchanger.

Figs. 3a and b are isometric views of a turbine/heat exchanger system in accordance with the present invention.

Fig. 4 is a perspective view of a cross-section of a portion of a heat exchanger in accordance with the present invention.

Fig. 5 is an angled cross-section of a portion of a heat exchanger in accordance with the present invention.

Figs. 6a and b are perspective views of cross-sections of a portion of a heat exchanger in accordance with the present invention.

Fig. 7 is a perspective view of a cross-section of a portion of a heat exchanger in accordance with the present invention.

Fig. 8 is a perspective view of a cross-section of a portion of a heat exchanger in accordance with the present invention.

Fig. 9 is a perspective view of a cross-section of a portion of a heat exchanger in accordance with the present invention.

Fig. 10 is a perspective view of a cross-section of a portion of a heat exchanger in accordance with the present invention.

Fig. 11 is a perspective view of a cross-section of a portion of a heat exchanger in accordance with the present invention.

Fig. 12 is a perspective view of a cross-section of a portion of a heat exchanger in accordance with the present invention.

DETAILED DESCRIPTION

The present invention is embodied in an apparatus which provides several advantages over prior devices. One such advantage is that the invention allows differential expansion or contraction between the core and the support structure without structural damage. Another advantage is that the airflow to, or from, the core is kept substantially unrestricted during the expansion and contraction of the core.

In at least some embodiments of the invention, the core is secured to the support structure at a single location and is allowed to expand out from the location and contract in towards it. It is preferred that the securing location is set near (e.g. adjacent) the core's higher temperature fluid port. The core can be secured to the support structure by a pin and receiver apparatus. A flexible connector is used to maintain fluid flow through the core during the core's expansion and contraction. It is preferred that this connector (e.g. bellows, flexible hose, etc.) is positioned at the core's lower temperature fluid port. This allows the connector to be designed and fabricated to transport only lower temperature fluids, reducing cost and complexity of the connector.

With the core held in place near the higher temperature fluid port, the rest of the core is free to expand and contract. As such, the lower temperature fluid port and the flexible connector move with the expansion and contraction of the core. While the flexible connector moves, it functions to maintain a substantially unrestricted fluid passage way between the core and any external structure (e.g. ducting) attached thereto.

Another advantage of the present invention is that, by allowing relatively free differential expansion and contraction of the core, it prevents damage which would otherwise occur by restricting the movement of the structures. This damage potentially would occur from a variety of sources including buckling, fatigue, creep or the like. Preventing such damage results in an increased life span of the heat exchanger and reduces the amount of supporting structure needed.

Still another advantage of the present invention is that the overall cost and complexity of the heat exchanger is reduced. This reduction is due to, among other things, the simplicity of construction, reduction in the structural elements and reduced material costs. For example, with the core secured at or near the core's higher temperature fluid port, a direct connection can be made from this port to any external structure (e.g. ducting), eliminating the need for a flexible connector at this location. Since a flexible connector at the higher temperature port must be able to withstand the extreme heat, while remaining sufficiently flexible, it must be made of relatively expensive materials. As such, the overall cost of the heat exchanger can be reduced. Further, eliminating this high temperature flexible output connector reduces the complexity of the heat exchanger, which in turn eases the assembly.

Heat exchanger apparatuses which provide for differential thermal expansion are set forth in U.S. Patent Application (Number to be assigned) filed on February 5, 2002, entitled HEAT EXCHANGER HAVING VARIABLE THICKNESS TIE RODS AND METHODS OF FABRICATION THEREOF, by David Beddome, Steve Ayres and Yuhung Edward Yeh, which is hereby incorporated by reference in its entirety, U.S. Patent Application (Number to be assigned), filed December 21, 2001, entitled HEAT EXCHANGER WITH BIASED AND EXPANDABLE CORE SUPPORT STRUCTURE, by David Beddome, Steve Ayres and Yuhung Edward Yeh, which is hereby incorporated by reference in its entirety, U.S. Patent Application No. 09/652,949, filed on August 31, 2000, entitled HEAT EXCHANGER WITH BYPASS SEAL ALLOWING DIFFERENTIAL THERMAL EXPANSION, by Yuhung Edward Yeh, Steve Ayres and David Beddome, which is hereby incorporated by reference in its entirety, and U.S. Patent Application No. 09/864,581, filed on May 24, 2001, entitled HEAT EXCHANGER WITH MANIFOLD TUBES FOR STIFFENING AND LOAD BEARING, by David W. Beddome, Steve Ayres, Yuhung Edward Yeh, Ahmed Hammoud, David Bridgnell and Brian Comiskey, which is hereby incorporated by reference in its entirety.

As shown in Figure 3a (Fig. 3a), for some embodiments, the present invention is a heat exchanger 100 which can be used in conjunction with a gas turbine engine. The heat exchanger 100 functions to heat the inlet fluid, in this case air, prior to it entering the turbine and cool the fluid exiting the turbine, in this case exhaust gases, prior to it exiting the heat exchanger 100. This is achieved by directing the inlet air so that it passes adjacent to the exhaust gas, such that heat is transferred from the exhaust to the inlet air. Specifically, as set

forth in Figure 3a, air enters at an air inlet and is directed through the heat exchanger 100 where it is heated by heat from the exhaust gases. Then, the heated air is directed from the heat exchanger 100 to the turbine. The turbine uses the air to operate and in so doing expels the exhaust gas. The exhaust gas is directed into and through the heat exchanger 100 where it heats the inlet air. The cooled exhaust gas then exits from the heat exchanger 100. A detailed description of the functioning and structure of the heat exchanger 100 is set forth herein.

While Figure 3a shows an example of a system in that some embodiments of the present invention are used, many other systems and uses are possible, including the use of engines other than a gas turbine, and fluids other than air and exhaust gases. In some embodiments of the present invention (as detailed below), the heat exchanger intakes a higher temperature fluid at its inlet and outputs a lower temperature fluid at its outlet.

Figure 3b (Fig. 3b) shows an embodiment of the heat exchanger 100 with an lower temperature fluid duct or air inlet 113 and a higher temperature fluid duct or air outlet 119, to bring air into and out of a heat transfer core (not shown), and an exhaust gas inlet and an exhaust gas outlet, to direct the exhaust gases through the heat exchanger 100. The heat exchanger 100 also has a shell assembly 160a with a first or upper strongback 143a and a second or lower strongback 145 (not shown) on either end. Connecting the strongbacks are a set of tie rods 150. Set between the air inlet and the core is a flexible connector 180a. Figure 3b sets forth the cross-sections of the heat exchanger 100 as shown in Figures 4 (Fig. 4) and Figure 5 (Fig. 5).

For some embodiments of the present invention, as shown in the cut-away views of Figures 4 and 5, the heat exchanger 100, has a core 110a positioned within the shell assembly 160a. Outside the shell 160a are the upper strongback 143a and the lower strongback 145, connected by the tie rods 150. The upper strongback 143a, the lower strongback 145, the tie rods 150, and the shell 160a, collectively form a support structure 170a. Positioned between the core 110a and the support structure 170a is a mount 200a. The flexible connector or bellows 180a is positioned between the air inlet 113 and a lower temperature fluid manifold tube or inlet manifold tube 115.

The core 110a is positioned within the shell 160a. The core 110a functions to duct the

inlet air pass the exhaust gas, so that the heat of the exhaust gas can be transferred to the cooler inlet air. The core 110a performs this function while keeping the inlet air separated from the exhaust gas, such that there is no mixing of the air and the gas. By moving air near the gas without mixing the two, the heat exchanger 100 transfers heat at a high level of efficiency. Further, the heat exchanger 100 also maximizes engine performance by not allowing the exhaust gases to be introduced into the intake air of the turbine (or other engine).

As shown in Figures 4 and 5, the core 110a has an exterior surface 112. A lower temperature fluid port, an air inlet port or first port 114 brings air into the core 110a and a higher temperature fluid port, air outlet port or second port 118 brings air out of the core 110a. The air inlet port 114 receives relatively cool inlet air for passage through the core 110a. When the heat exchanger 100 is operating, the air exiting the air outlet port 118, having been heated in the core 110a, will have a much higher temperature than the inlet air. Between the air inlet port 114 and the air outlet port 118 are the inlet manifold tube 115, a lower temperature fluid manifold or inlet manifold 116, a heat exchange region 122, a higher temperature fluid manifold or outlet manifold tube 117, and a higher temperature fluid manifold or outlet manifold 120.

While the heat exchanger 100 is operating, the core 110a has a variable size (e.g. length and width) caused by thermal expansion or contraction. That is, as the core 110a is heated up by the exhaust gases passing through the shell, the core 110a will expand and as the heat exchanger 100 stops operating the core 110a will contract as it cools.

The heat exchange region 122 can be any of a variety of configurations that allow heat to transfer from the exhaust gas to the inlet air, while keeping the gases separate. However, it is preferred that the heat exchange region 122 is a prime surface heat exchanger having a series of layered plates 128, which form a stack 130. The plates 128 are arranged to define heat exchange members or layers 132 and 136 which alternate from ducting air, in the air layers 132, to ducting exhaust gases, in the exhaust layers 136. These layers typically alternate in the core 110a (e.g. air layer 132, gas layer 136, air layer 132, gas layer 136, etc.). Separating each layer 132 and 136 is a plate 128.

On either end of the stack 130 are a first end plate 142a and a second end plate 144. The first end plate 142a is positioned against the upper portion of the shell assembly 160a and

the second end plate 144 is positioned against the lower portion of the shell assembly 160a.

Also shown in Figure 4, are the tie rods 150 positioned on either side of the core 110a. A series of the tie rods 150 and an upper strongback or load bearing member 143a and a lower strongback or load bearing member 145, are used to hold the stack 130 together and carry loads. The tie rods 150 function to apply a compressive load to the strongbacks 143a and 145. The tie rods 150 include a center section 151 running between either end 152 and fasteners 153 at each end 152. The fasteners 153 function to hold the tie rods 150 to the strongbacks 143a and 145. The tie rods 150 can be made of any suitable well known material including, but not limited to, steel and aluminum. However, the tie rods 150 are preferably stainless steel. The tie rods 150 are described in further detail below.

On the outside of the shell 160a and above and below the core 110a, are the upper strongback 143a and the lower strongback 145. The tie rods 150 and the strongbacks 143a and 145 (as well as the shell 160) carry compressive loads applied to the stack 130. These compressive loads can be from a variety of sources including pre-loading, differential thermal expansion, air pressure, and the like.

The upper strongback 143a, the lower strongback 145, the tie rods 150, and the shell 160a, together form the support structure 170a. The support structure 170a functions to apply the compressive force to the stack 130 of the core 110a. In contrast to the tie rods 150, the upper strongback 143a and the lower strongback 145 are generally not deformable.

As can be seen, the plates 128 are generally aligned with the flow of the exhaust gas through the shell assembly 160a. The plates 128 can be made of any well known suitable material, such as steel, stainless steel or aluminum, with the specific material dependent on the operating temperatures and conditions of the particular use. The plates 128 are stacked and connected (e.g. welded or brazed) together in an arrangement such that the air layers 132 are closed at their ends 134. With the air layers 132 closed at ends 134, the core 110a retains the air as it passes through the core 110a. The air layers 132 are, however, open at air layer intakes 124 and air layer outputs 126. As shown in Figures 5 and 6, the air layer intakes 124 are in communication with the inlet manifold 116, so that air can flow from the inlet manifold tube 115 through the inlet manifold 116 and into each air layer 132. Likewise, the air layer outputs 126 are in communication with the outlet manifold 120, to allow heated air to flow

from the air layers 132 through the outlet manifold 120 and out the outlet manifold tube 117.

In contrast to the air layers 132, the gas layers 136 of the stack 130 are open on each end 138 to allow exhaust gases to flow through the core 110a. Further, the gas layers 136 have closed or sealed regions 140 located where the layers 136 meet both the inlet manifold 116 and the outlet manifold 120. These closed regions 140 prevent air, from either the inlet manifold 116 or the outlet manifold 120, from leaking out of the core 110a into the gas layers 136. Also, the closed regions keep the exhaust gases from mixing with the air.

Therefore, as shown in Figures 4 and 5, the intake air is preferably brought into the core 110a via the inlet manifold 116 and distributed along the stack 130, passed through the series of air layer intakes 124 into the air layers 132, then sent through the air layers 132 (such that the air flows adjacent - separated by plates 128 - to the flow of the exhaust gas in the gas layers 136), exited out of the air layer 132 at the air layer outputs 126 into the outlet manifold 120, and finally out of the core 110a. In so doing, as the air passes through the core 110a, it receives heat from the exhaust gas.

With the stack 130 arranged as shown in Figures 4 and 5, the hot exhaust gas passes through the core 110a at each of the gas layers 136. The exhaust gas heats the plates 128 positioned at the top and bottom of each gas layer 136. The heated plates 128 then, on their opposite sides, heat the air passing through the air layers 132.

As the plates 128 and the connected structure of the core 110a heat up, they expand. This results in an expansion of the entire stack 130 and thus of the core 110a. As noted in detail below, the inlet bellows 180a and the mount or restraining apparatus 200a are configured to allow the core 110a to thermally expand separately from the support structure 170a. In this manner, the core 110a can expand and contract laterally without the build-up of excessive forces between the core 110a and the support structure 170a and without the use of a bellows at the air outlet port 118 of the core 110a. This saves the core 110a from being damaged by forces which would otherwise be created by affixing the core 110a in place. Also, it reduces the cost of the heat exchanger 100 by eliminating the need for an expensive outlet bellows.

Although the core 110a can be arranged to allow the air to flow through it in any of a variety of ways, it is preferred that the air is channeled so that it generally flows in a direction opposite, or counter, to that of the flow of the exhaust gas in the gas layers 136 (as shown in the cross-section of Figure 4). With the air flowing in an opposite direction to the direction of the flow of the exhaust gas, it has been found by the Applicants that the efficiency of the heat exchanger is significantly increased as compared to other flow configurations. As noted in detail below, some embodiments of the present invention have the core functioning to cool hot fluid entering the core inlet with a cooler fluid being direct through the shell.

The arrangement of the core 110a can be any of a variety of alternate configurations. For example, the air layers 132 and gas layers 136 do not have to be in alternating layers, instead they can be in any arrangement which allows for the exchange of heat between the two layers. For example, the air layers 132 can be defined by a series of tubes or ducts running between the inlet manifold 116 and the outlet manifold 120. While the gas layers 136 are defined by the space outside of, or about, these tubes or ducts.

To facilitate heat transfer, the core 110a can also include secondary surfaces such as fins or thin plates connected to the inlet air side of the plates 128 and/or to the exhaust gas side of the plates 128.

The core 110a and shell 160a can carry various gases, other than, or in addition to, those mentioned above. Also, the core 100a and shell 160a can carry any of a variety of fluids.

As shown in Figures 4 and 5, the shell assembly 160a includes side walls 162, openings 164, an upper panel 166a and a lower panel 168. The shell assembly 160a functions to receive the hot exhaust gases, channel them through the core 110a, and eventually direct them out of the shell 160a. The shell 160a is relatively air tight to prevent the exhaust gases from leaking out of the shell 160a. The shell 160a is large enough to fully contain the core 110a and at least strong enough to withstand the pressure exerted on the shell 160a by the exhaust gas. Typically, the shell 160a is flexible and can be deformed to varying amounts depending on its specific construction.

The openings 164 of shell 160a are positioned through the upper panel 166a. The shell assembly 160a can be made of any suitable well known material including, but not

limited to, steel and aluminum. Preferably, the shell 160a is a stainless steel.

5 The construction of the shell assembly 160a can vary depending on the particular embodiment of the present invention. In some embodiments the shell 160a is constructed to carry some of the compressive load generated by the support structure 170a and applied to the core 110a. The shell 160a can also be configured to carry other internally created loads (e.g. air pressure loads) and externally exerted loads (e.g. inertia loads or vibration loads). Because in some embodiments of the present invention, the walls 162, upper panel 166a and lower panel 168 of the shell 160a are thick relative to the thin core plates 128, the shell 160a will thermally expand at a slower rate than the core 110a. This can result in differential thermal expansion or contraction between the shell 160a and the core 110a, as the two are either heated or cooled, as the case may be. To avoid, or to minimize, gaps or spaces forming between the core 110a and the shell 160a during differential expansion, the shell 160a is flexible enough to be deformed by the forces applied by the strongbacks 143a and 145 and the tie rods 150.

15 In other embodiments, the structure of the shell 160a is relatively thin. In such embodiments, the compressive loads created by the support structure 170a are primarily carried by the strongbacks 143a and 145 and the tie rods 150. In such embodiments, because the shell 160a is thinner, the shell 160a, thermally expands and contracts much quicker. This allows any differential thermal expansion between the shell 160a and the core 110a to be minimized. Which, in turn, aids in preventing gaps from forming between the core 110a and the shell 160a. This thinner structure also increases the shell's flexibility and allows the shell 160a to be more easily deformed by the strongbacks 143a and 145 and the tie rods 150. As such, in these embodiments, the potential for exhaust gases being able to pass around the core 110a, through gaps between the core 110a and the shell 160a, is further reduced.

20 The present invention, however, provides for differential thermal expansion between the structures of the heat exchanger 100 by employing the inlet bellows 180a and the mount 200a to allow the core 110a to thermally expand separately from the support structure 170a, while maintaining a substantially unrestricted airflow through the core 110a. As shown herein, a variety of embodiments of the support structure and tie rods exist.

As shown in Figures 4 and 5, one embodiment of the present invention has the core 110a fixed to the support structure 170a near the air outlet port 118 and movable near the air inlet port 114. This embodiment allows the core 110a to expand and contract freely in a lateral direction, while preventing damage to components of the heat exchanger 100 and while maintaining a sealed and unobstructed flow of air through the core 110a.

This embodiment is achieved by using the deformable connector, flexible bellows or hose 180a positioned between the air inlet port 114 and any external air ducting (e.g. the air inlet). A direct, substantially rigid or fixed outlet connector 190a is set between the air outlet port 118 and any external ducting (e.g. the air outlet).

With the core 110a fixed in place by the mount 200a near air outlet port 118 and the outlet connector 190a, the core 110a will have little or no movement at the outlet connector 190a during the differential thermal expansion or contraction of the core 110a. All the lateral expansion and contraction of the core 110a occurs out away from the mount 200a, and thus, out from the connector 190a (being positioned in close proximity to the mount 200a).

As such, the outlet connector 190a can be fixed and does not have to deform (at least not in any significant manner) to accommodate the differential expansion or contraction of the core 110a. That is, as shown in Figures 4 and 5, the outlet connector 190a is a straight section extending between the air outlet port 118 and the air outlet duct. The connector 190a can be any of a variety of shapes and/or lengths, however it is preferred that the connector 190a is shaped to match the shapes of the outlet 118 and the outlet manifold 120. Specifically, it is preferred that the connector 190a is a tube with a round cross-section. The connector 190a is preferably stainless steel, but other materials including steel and aluminum can be used.

It should be noted that since the mount 200a is slightly offset from the connector 190a that in some embodiments of the present invention the connector 190a may be subject to some relatively minor lateral deformation. It is preferred that the connector 190a be sufficiently laterally deformable to accept any such differential expansion. As such, a bellows is not needed between the air outlet port 118 the air outlet duct.

By not needing to use an outlet bellows, the present invention reduces the cost and complexity of the heat exchanger 100. A bellows set between the manifold 120 and the outlet 118 would have to remain sufficiently flexible at the higher temperatures found at the core's outlet. Such bellows are significantly more expensive and complex than a straight connector, such as the outlet connector 190a.

Of course, because the air inlet port 114 is positioned much further away from the mount 200a than the air outlet port 118, the lateral movement of the core 110a is much greater at the air inlet port 114 than at the air outlet port 118. To maintain a sealed and generally clear path for the inlet air, the inlet bellows 180a is positioned between the air inlet port 114 and the air inlet duct, as shown in Figures 4 and 5.

As shown in Figures 6a and b (Figs. 6a and b), the inlet bellows 180a includes a lower portion 182a, an upper portion 184a and side walls 186a. The lower portion 182a is mounted to the core 110a at the air inlet port 114. The upper portion 184a is mounted to the external air inlet duct. The side walls 186a are deformable both laterally and along the length of the bellows 180a. The side walls 186a include alternating planar sections 188a.

The inlet bellows 180a can be any of a variety of materials including steel and aluminum, however it is preferred that stainless steel is used. In place of a bellows a flexible high temperature hose or a braided (e.g. woven) metal hose can be used.

The bellows 180a can be any of a variety of shapes and dimensions, however, it is preferred that the bellows 180a have a round shape to match that of the preferred tube shapes of the air inlet port 114 and air inlet tube. The length of the bellows 180a can vary, but is preferably dependent on the maximum differential expansion and/or contraction of the core 110a. The greater the overall difference between the lateral dimensions of the core 110a and the support structure 170a, the greater length of the bellows 180a will be.

As the core 110a expands or contracts, the inlet manifold 116 moves laterally to one side or the other, relative to the support structure 170a, as shown in Figures 6a and b. As the inlet manifold 116 moves laterally, it carries along with it the inlet manifold tube 115. The inlet bellows 180a deforms laterally to allow air to flow from the air inlet duct through the bellows 180a and into the inlet manifold tube 115 (via the air inlet port 114). Figure 6a shows the core 110a having differentially expanded laterally away from the mount 200a

5 faster than the lateral expansion of the support structure 170a. As a result, the bellows 180a has shifted its lower portion 182a to the left with the inlet manifold tube 115. The inlet manifold tube 115 moves within an expansion opening 111 formed in the upper strong back 143a. The expansion opening 111 is sized and shaped to allow the inlet manifold tube 115 to move without contact with the upper strong back 143a. The specific size of the expansion opening 111 can vary and is dependent on the maximum amount of differential expansion and contraction of the core 110a.

10 In contrast, Figure 6b shows the core 110a having contracted towards the mount 200a quicker than the support structure 170a. In so doing, the bellows 180a has had its lower portion 182a shifted to the right relative to the upper portion 184a. The inlet manifold tube 115 has moved to the right in the expansion opening 111.

15 In either the case of the differential expansion or contraction of core 110a, the inlet bellows 180a maintains a seal with the inlet manifold tube 115 and with air inlet duct. As can be seen, with either the core's expansion or contraction, the bellows 180a maintains a clear pathway for the passage of air into the core 110a.

20 As can be seen in Figures 4 and 5, the mount 200a is positioned between the core 110a and the support structure 170a. It is preferred that the mount 200a is positioned near the air outlet port 118, such that any movement of the core 110a relative to the support structure 170a at the connector 190a is minimized. This eliminates the need for a separate bellows to be used between the air outlet port 118 and the air outlet duct, resulting in a reduction of the overall cost and complexity of the heat exchanger 100.

25 Of course, the mount 200a can be positioned at any of a variety of locations about the connector 190a other than that shown in Figures 4 and 5. While it is preferred that the mount 200a is kept relatively close to the connector 190a, depending on the specific amount of maximum differential expansion and contraction, the position of mount 200a relative to the connector 190a can vary. That is, generally the less the differential expansion and contraction, the further the mount 200a can be positioned laterally away from the connector 190a without overly deforming the connector 190a or damaging it.

As shown in Figure 7 (Fig. 7), the mount 200a includes the pin 202a and a receiver, mating relief or recess 206a. In at least one embodiment, the pin 202a is attached to the upper strongback 143a of the support structure 170a and extends towards the core 110a. The pin 202a includes sides 204a. The pin is received in a receiver 206a, which in this embodiment, is a hole defined in the first end plate 142a. The receiver 206a includes sides 208a.

Figures 6a and b and 7 show that the pin 202a is positioned in the receiver 206a, such that the pin 202a restrains movement of the core 110a relative to the support structure 170a at the mount 200a. As the core 110a begins to displace laterally, the sides 204a of the pin 202a contact the sides 208a of the receiver 206a to prevent the core 110a from moving. However, since the remainder of the core 110a can move laterally substantially freely (with the first end plate 142a moving adjacent to the upper panel 166a of the shell 160), the core 110a will expand out from the mount 200a and contract towards it. As such, the expansion and/or contraction at the connector 190a will be much less than that at the bellows 180a.

The pin 202a can be any of a variety of materials including steel and aluminum but it is preferred that stainless steel is used. The pin 202a preferably has a cylindrical shape, of course other shapes are possible as well.

The pin 202a is secured to the upper strongback 143a and for additional strength can also be secured to the shell 160a. In some embodiments, the pin 202a is attached to the shell 160a and/or the upper strongback 143a by welding, brazing, adhesives or any similar method. In other embodiments, the pin 202a can be a formed part of either the strongback 143a (as shown in Figures 4-7) or the shell 160a. In at least some embodiments the pin 202a is a tab which is bent, or otherwise deformed, from the shell 160a and/or the upper strongback 143a.

The dimensions of the pin 202a are variable, depending on the specific use in which it is employed and material used. The dimensions of the pin can be determined by one skilled in the art using well known analytical and/or empirical methods.

The receiver 206a can be created by forming, drilling and/or any other similar well known method. The receiver 206a is sized to closely receive the pin 202a. This prevents lateral movement of the core 110a at the mount 200a.

5 The mount 200a, including the pin 202a and the receiver 206a must be strong enough to carry the loads generated by the differential thermal expansion and/or contraction of the core 110a, without any significant damage to the mount 200a. The mount 200a needs to be able to carry such loads over repeated cycles of differential expansion and contraction of the core 110a.

10 Certain embodiments of the present invention use more than one mount 200a to secure the core 110a. It is preferred that such embodiments have the mounts 200a positioned close enough to each other to prevent damage from differential expansion and/or contraction of the core 110a. In certain embodiments the multiple mounts 200a are positioned about the outlet manifold 120 so as to minimize or prevent lateral movement of the core 110a at the connector 190a.

15 In some embodiments of the present invention, the mount has a reverse arrangement. As shown, in Figure 8 (Fig. 8), a mount 200b has a pin 202b which is attached to the core 110b and which extends into a receiver 206b positioned in the support structure 170b. The pin 202b is secured to first end plate 142b. The pin 202b can be a formed part of the first end plate 142b (as shown) or it can be attached thereto by welding, brazing, adhesives or any similar method. In other embodiments the pin 202b is a tab which is material bent out from the first end plate 142b. The receiver 206b is defined out of the upper panel 166b of the shell 160b and/or out of the upper strongback 143b. The receiver 206b can be created by forming, drilling and/or any other similar well known method.

20 In other embodiments, a mount 200c includes a pin 202c, a core receiver 206c and a support structure receiver 207c, as shown in Figure 9 (Fig. 9). The pin 202c is received by both the core receiver 206c and the support structure receiver 207c. In this manner, the core 110c is held in place by the pin 202c being held laterally by both the receiver 206c and the receiver 207c. In these embodiments the core receiver 206c and the support structure receiver 207c are defined by forming, drilling or the like.

30 As shown in Figure 10 (Fig. 10), in another embodiment of the present invention, a mount 200d is positioned about the outlet manifold 120. In this embodiment the mount includes a ring 203d and a receiver 206d. The ring 203d is attached to the support structure

170d about the manifold tube 117d and extends into the receiver 206d. The receiver 206d is defined in the core 110d about the outlet manifold 120. The mount 200d allows the core to laterally expand about the outlet manifold 120 while preventing lateral movement at the outlet manifold 120. This provides the benefit that the connector 190d is not deformed during the differential expansion and contraction of the core 110d. This embodiment continues to use a bellows (not shown) between the air inlet (not shown) and the inlet manifold tube (not shown). As with other embodiments of the present invention (as detailed above), the mount 200d can have a variety of embodiments. The mount 200d can have a ring mounted to the core 110d and a receiver defined in the supporting structure 170d, or both the core 110d and the support structure 170d can have a receiver, which has each receiver accepting a portion of a ring set therebetween. Also, the mount 200d can be a set of pins and receivers positioned about the outlet manifold 120. In some embodiments of the present invention, the ring 203d and the connector 190d are attached or formed as a single structural element.

In still other embodiments of the present invention, the core 110e and the support structure 170e are attached in a fixed manner to one another. As shown in Figure 11 (Fig. 11), a mount 200e includes a pin or tab 202e extending from the first end plate 142e to the support structure 170e. The pin 202e is secured to the shell 160e and the upper strongback 143e. The pin 202e can be secured by any of a variety of methods including by welding, brazing, the use of adhesives, or the like. The weld, brazing or adhesive 205e secures the pin 202e to the shell 160e and the upper strongback 143e. The pin 202e can be formed from the first end plate 142e (as shown) or otherwise affixed thereto. In other embodiments the pin 202e can extend from the support structure 170e and be welded, brazed or otherwise adhered to the core 110e. The pin 202e can also be set between the support structure 170e and the core 110e and welded, brazed or otherwise adhered to both the support structure 170e and the core 110e.

In some embodiments of the present invention a bellows 180f is positioned between the core 110f and the support structure 170f, as shown in Figure 12 (Fig. 12). In this position as the core 110f expands away from and contracts towards the mount 200f, the bellows 180f maintains a substantially unobstructed fluid pathway between the air inlet and the core 110f. The first end plate 142f is positioned away from the bellows 180f (about the bellows) to provide space for the bellows 180f as the first end plate 142f moves with the expansion and